

GUJARAT TECHNOLOGICAL UNIVERSITY**BE- VIth SEMESTER-EXAMINATION – MAY- 2012****Subject code: 160101****Date: 09/05/2012****Subject Name: Aerodynamics II****Time: 10:30 am – 01:00 pm****Total Marks: 70****Instructions:**

1. Attempt all questions.
2. Make suitable assumptions wherever necessary.
3. Figures to the right indicate full marks.

Q.1 (a) Explain critical Mach number, drag divergence Mach number, sound barrier, Area rule and super critical airfoil. Is there any relation among the parameters or phenomenon mention? Explain. **07**

(b) Explain profile theory for two dimensional flap wing **07**

Q.2 (a) Explain compressibility effect on fuselage system **07**

(b) State and Explain Kutta condition with the help of diagram. Also explain importance of Kutta condition **07**

OR

(b) Explain low speed flow over an airfoil **07**

Q.3 (a) Using mathematical derivation provide the solution of generalized equation for symmetric airfoil using thin airfoil theory **07**

(b) Explain steps for exact Joukowski transformation process and its numerical solution **07**

OR

Q.3 (a) Explain in details with procedure about transformation of circle into symmetric airfoil **07**

(b) Write notes on bound vortex, Horse shoe vortex system **07**

Q.4 (a) Explain Biot-Savart Law for vortex and Derive equation for infinite wing **07**

(b) State Helmholtz's laws for vortex. Also explain how this laws are useful in finite wing theory? **07**

OR

Q.4 (a) Write note on finite airfoil theory **07**

(b) Explain and derive suitable mathematical equation for minimum drag load distribution on a given wing. **07**

Q.5 (a) Explain application of linearized to supersonic airfoil **07**

(b) Write note on aerodynamics of fuselage system **07**

OR

Q.5 (a) Write notes on aerodynamics of airfoil, infinite wing and finite wing **07**

(b) Write note on viscous flow past an airfoil **07**
