

**GUJARAT TECHNOLOGICAL UNIVERSITY****B. E. - SEMESTER – VII • EXAMINATION – WINTER 2012****Subject code: 170106****Date: 28/12/2012****Subject Name: Viscous flow and Boundary Layer Theory****Time: 10.30 am - 01.00 pm****Total Marks: 70****Instructions:**

1. Attempt any five questions.
2. Make suitable assumptions wherever necessary.
3. Figures to the right indicate full marks.

- Q.1 (a)** Define boundary layer thickness, momentum thickness and energy thickness, and derive expression for the energy thickness for laminar flow over a flat plate. **07**
- (b)** Write a short note on boundary layer control methods. **07**

- Q.2 (a)** Derive the von-Karman momentum integral equation for the flow past a flat plate in the form **07**

$$\tau_0 = \rho U_\infty^2 \frac{\partial \theta}{\partial x}$$

Where,  $\theta$  is momentum thickness

- (b)** Derive Orr-Sommerfeld Equation. **07**

**OR**

- (b)** What is critical Reynolds number? State OSE, discuss the solution of it and show how critical Reynolds number is obtained from that. And derive Rayleigh equation from the OSE. **07**

- Q.3 (a)** Explain Prandtl's mixing length hypothesis and derive expression for mixing length. **07**
- (b)** Write a short note on Relaminarization. **07**

**OR**

- Q.3 (a)** Write down the characteristics of the turbulent flow. **07**
- (b)** Explain the evolution of the turbulent flow from laminar flow. **07**

- Q.4 (a)** Derive the expression to determine boundary layer thickness in turbulent boundary layer. **07**

- (b)** Air is flowing over a flat plate 5m long and 2.5m wide with a velocity of 4m/s at 15°C. If  $\rho = 1.208 \text{ kg/m}^3$  and  $\nu = 1.47 \times 10^{-5} \text{ m}^2/\text{s}$ . calculate **07**
- 1) Length of plate over which the boundary layer is laminar, and thickness of the boundary layer at that point
  - 2) Shear stress at the location where boundary layer ceases to be laminar.
  - 3) Total drag force on the both sides on that portion of plate where boundary layer is laminar.

**OR**

- Q.4 (a)** Derive energy equation for thermal boundary layer in the form of **07**

$$u \frac{\partial t}{\partial x} + v \frac{\partial t}{\partial y} = \frac{k}{\rho c_p} \frac{\partial^2 t}{\partial y^2} + \frac{\mu}{\rho c_p} \left( \frac{\partial u}{\partial y} \right)^2$$

- (b)** A flat plate 1m wide and 1.5 m long is maintained at 90°C in free stream of air having 10°C. Determine the velocity with which air must flow over a flat **07**

plate such that the rate of energy transfer from plate is 3.75kW. properties of air are  $\rho=1.09\text{kg/m}^3$ ,  $c_p=1.007\text{Ns/m}^2$ ,  $\mu=20\times 10^{-6}\text{kg/m}$  and  $\text{Pr}=0.7$

- Q.5 (a)** Derive momentum equation for turbulent flow in terms of Reynolds stresses. **07**
- (b)** Explain the flow development in laminar duct flow entering in a tube with uniform velocity of  $U_\infty$ , show velocity profiles at different length from entry. Relate entry length with height of duct. **07**

**OR**

- Q.5 (a)** Air at a temperature  $30^\circ\text{C}$  flows past a flat plate at a velocity of  $1.8\text{m/s}$ . the flat surface has a sharp leading edge and its total length equals  $750\text{mm}$ . calculate **07**
- 1) Average skin friction or drag co-efficient
  - 2) The average shear stress, and
  - 3) Ratio of average shear stress to the shear stress at the trailing edge.
- (properties of air at  $30^\circ\text{C}$  are,  $\rho=1.165\text{kg/m}^3$  and  $\nu=16\times 10^{-6}\text{m}^2/\text{s}$ )
- (b)** Find the ratio of friction drag on the front half and rear half of the flat plate **07**  
kept at zero incidence angle in a stream of uniform velocity, if the boundary layer is turbulent over whole plate.

\*\*\*\*\*